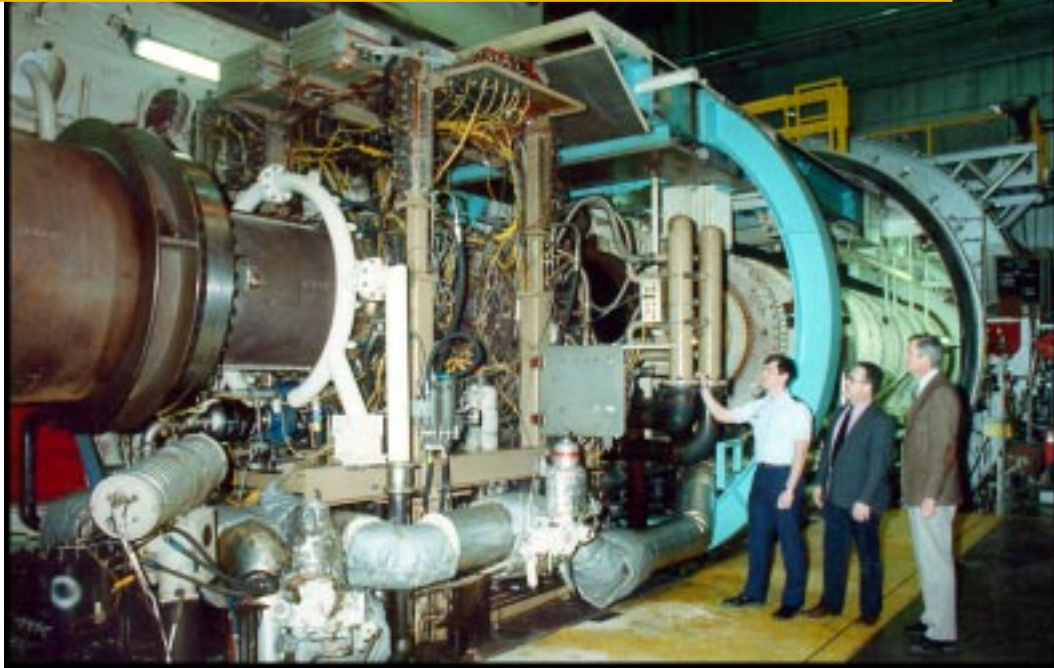


## ***ENGINE TEST FACILITY***

### ***Propulsion Development Test Cell J-1***



Engine Test Facility (ETF) - Propulsion Development Test Cell J-1 is located at Arnold Engineering Development Center, Arnold Air Force Base, Tennessee. The J-1 is an altitude test cell used to perform turbojet, turbofan, turboshaft, ramjet, aerodynamic models, or combined aerodynamic inlet and propulsion system tests. Fixed Mach Number Free Jet, or Direct-Connect tests of air-breathing propulsion systems to a simulated Mach number of 3.2 can be accomplished. It is primarily used for Direct-connect performance and stability testing of large air-breathing propulsion systems, although free-jet testing can be accommodated.

# PROPULSION DEVELOPMENT TEST CELL J-1

## Test Cell Dimensions:

Length: 65 ft  
Diameter: 16 ft  
Inlet Plenum Diameter: 8 ft

## Environmental Capabilities:

Altitude (ft): Sea Level to 80,000  
Total Pressure (psia): up to 120  
at the air inlet to the test cell  
Total Temperature (°F): -65 to 750  
Airflow (lb/sec): 0 to 1400  
Installed Thrust Stand Capacity (lbs):  
50,000 (on-stand calibration:  
32,000)

## Soak Capabilities:

Maximum Temperature (°F): 170  
Minimum Temperature (°F): -65  
The entire test cell is not brought  
to temperature, but instead, the  
engine is enclosed in a special set-  
up box (engine specific) that is  
brought to temperature. Soak  
duration is unlimited.

## Power Absorption:

J-1 has a 600 Hp extraction water  
brake that can be used in this cell.

## Data Processing Capabilities:

ETF data processing capabilities  
provide pre-test, test, and post-test  
data reduction and analysis.  
General programs are available for  
processing data acquired by  
the digital data acquisition system.  
These programs calculate  
calibration factors, convert raw  
data to engineering units, calculate  
performance analysis parameters,  
generate hard copy tabulations and  
plots, provide interactive  
alphanumeric and graphics  
displays, and supply inputs for  
special-purpose processing  
programs. New programs may be  
developed as needed to meet  
specific test-unique data reduction  
and analysis requirements. In

addition to real-time displays,  
data available for analysis / review  
during testing include all steady-  
state condition parameters and  
selected portions of time-  
dependent parameters. General  
data reduction programs are  
available for off-line processing of  
data recorded in the form of  
frequency-analog signals.

## Unique Features:

High accuracy measurement for  
multi-component thrust (axial,  
vertical side forces) is available.  
The test cell inlet airflow  
measuring system consists of five  
fixed and four remotely operated  
venturis.

*For more information on Engine Test  
Facility - Propulsion Development Test  
Cell J-1 at Arnold Engineering  
Development Center, Arnold Air Force  
Base, TN, contact 615-454-5851.*

## INSTRUMENTATION CAPABILITIES:

### Steady State

Number of Channels:  
Range:  
Sampling Rate

### Temperature

536+  
-300 to 3000°F/Varies  
-----20+ samples/sec/channel-----

### Pressure

512+  
5 to 600 psia

### Transient

Number of Channels:  
Range  
Sampling Rate:

### Temperature

536+  
-300 to 3000°F  
-----200+ samples/sec/channel-----

### Pressure

200+  
Varies

### Vibrations

24+  
Varies